

# **Inner Detector Cooling Review**

## **Pixels Evaporative Cooling Analysis May 1999**

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HYTEC**

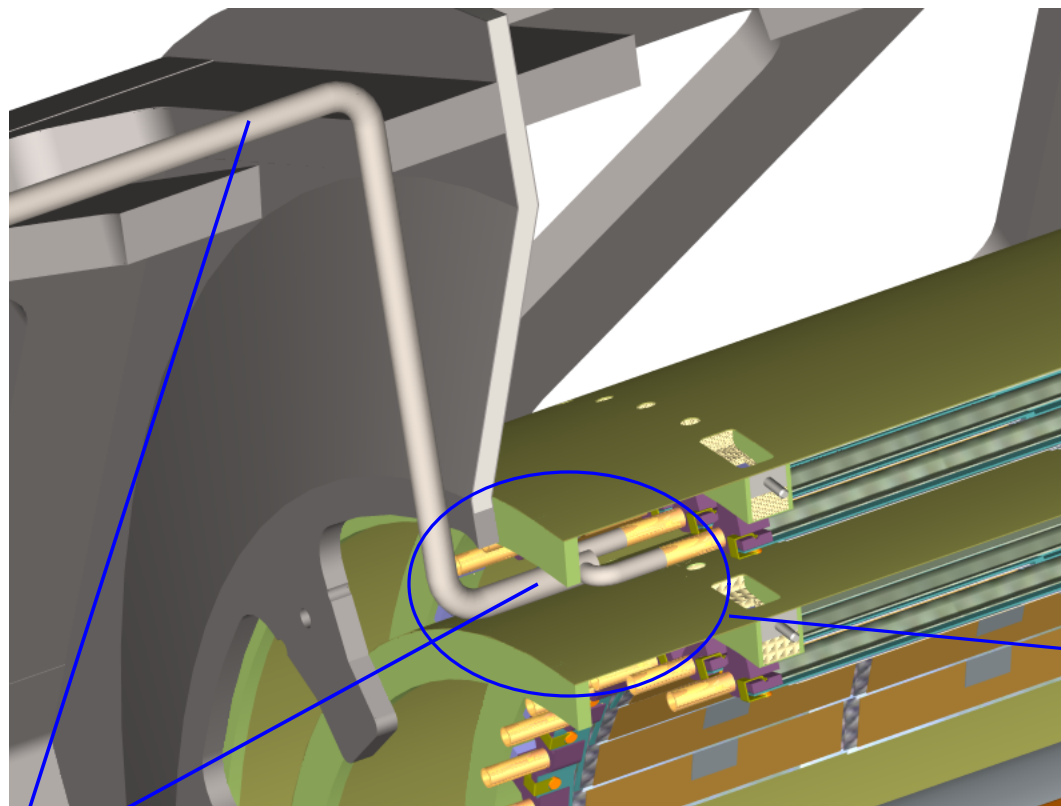
## Analysis of Vapor Return

- **Basic Approach**
  - **Low pressure  $C_4F_{10}$  evaporative cooling of detector modules**
    - concept demonstrated with extensive prototype testing
    - employs wet mixture of vapor achieved through throttling fluid at nominally 2 bar at entrance to carbon-carbon thermostructures
    - ~500 mbar wet mixture evaporates in thermostructure, exiting at quality on the order of 0.88(?)
    - low pressure in the thermostructures, as opposed to very high pressures:
      - limit distortion
      - minimize material
      - reduce risk
- **Questions to be addressed in this discussion**
  - *What impact does a low pressure system choice have on the vapor return line size*
  - *Would we be wise to seriously consider a condenser?*

## Analysis Scope

- Fluid calculations for vapor return-*first cut to verify low pressure concept*
  - establish minimum line size consistent with objective of providing 250 mbar at a compressor or a condenser inlet, depending upon concept
  - Consider potential flow states *and their effect on line losses*
    - single phase vapor-
      - isothermal versus adiabatic wall condition
      - minimizes system complexity and pressure gradients
      - potential incompressible flow solution for Mach number<0.3
    - two phase flow-
      - evaluate effect of quality (0.5 to >0.9) on pressure gradients in vapor return line
- System concept-*arrive at technical approach for low pressure system*
  - Condenser versus compressor concept
  - Thermodynamics of system operation and conservation of fluid inventory

## Stave Return Line Used As Example



Region of smallest tube size and highest velocity

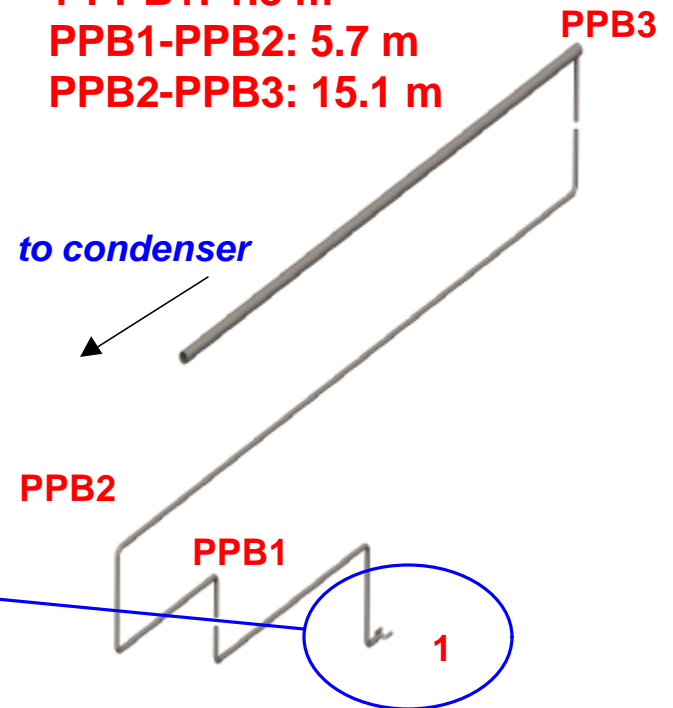
*Path lengths-new*

1-PPB1: 1.8 m

PPB1-PPB2: 5.7 m

PPB2-PPB3: 15.1 m

to condenser



## Evaluation Process

- **General approach**
  - first cut at tube sizes within the detector region where space constraints impose significant limitations on size
    - started with <6 mm initial tube ID
  - iterate with staff working service layout for larger tube, as required
  - make first cut at heat gain and tube outer surface temperatures
    - will require a number of iterations
  - evaluate prospect for achieving minimum 250 mbar return pressure out to 140 meters
    - presumed location of compressor
  - Based solution on 450 mbar exit pressure from stave
    - 200 mbar allowed if 250 mbar is to be realized at compressor inlet

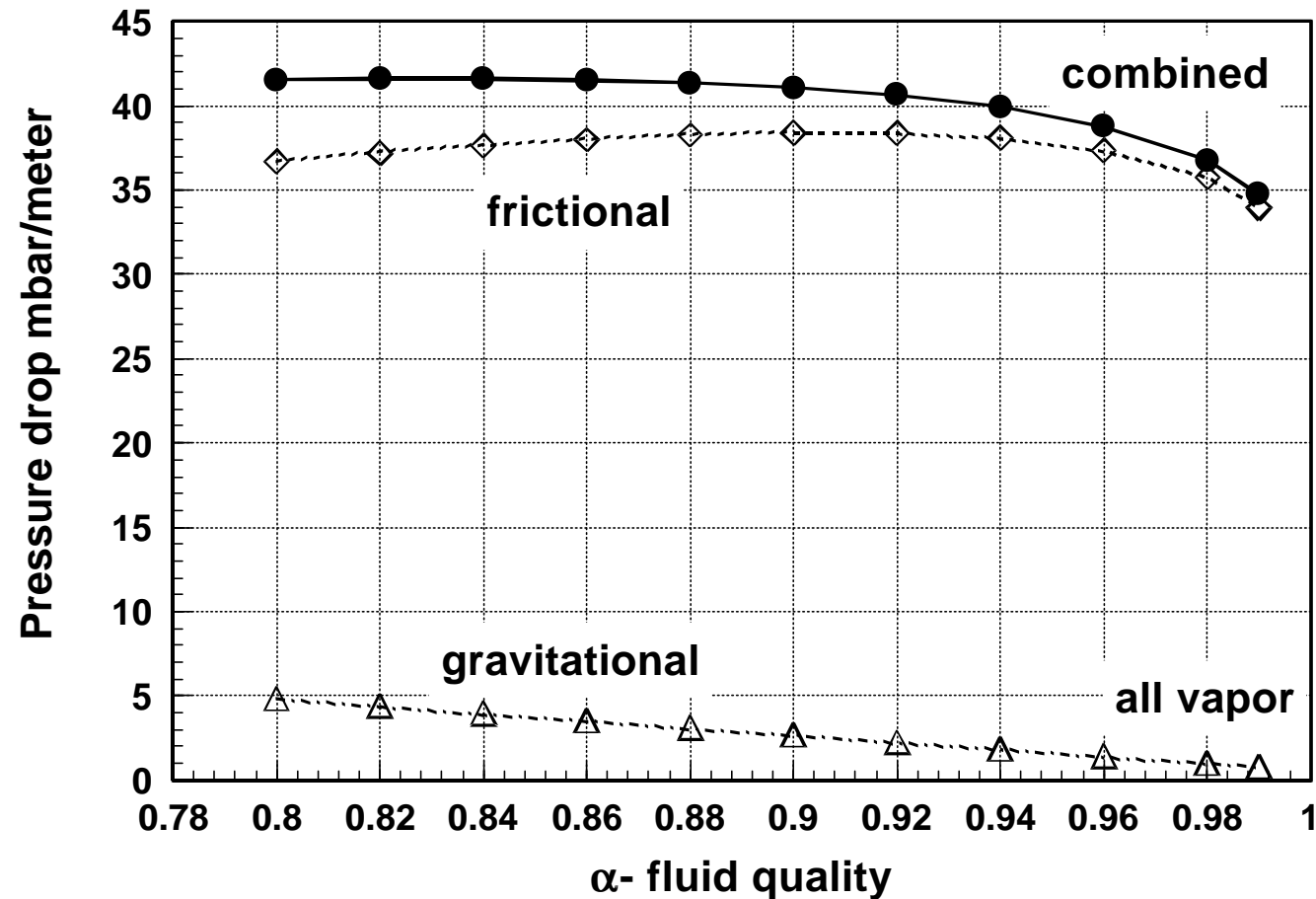
**Single Phase- Dry Vapor Return**  
**(6 mm diameter)**

Tube Section	Tube Diameter mm	Pressure Loss mbar	Static Exit Pressure mbar	Fluid Velocity m/sec	Fluid Density kg/m <sup>3</sup>	Remarks
@ Stave exit	3.4		450	41.9	5.1	Complete loss of dynamic head
Stave manifold Y-Branch	6 after Branch	8		16.8	5.01	Accounts for pressure loss merging into 6 mm tube
			441.7	16.8	5.01	Entrance to 1.5 m run
1.5 m	6	39.3				
	6	11.2				3 elbows
		50.5 combined	391.2	18.98	4.44	Exit after 1.5 m run
5.4	6	208.2				
		21.7				3 elbows
		229.9 combined	136.5	46.03	1.83	Exit after 5.4 m run
25	13	6.3	141.6	4.7	1.61	Exit after 25 m run, with some pressure recovery



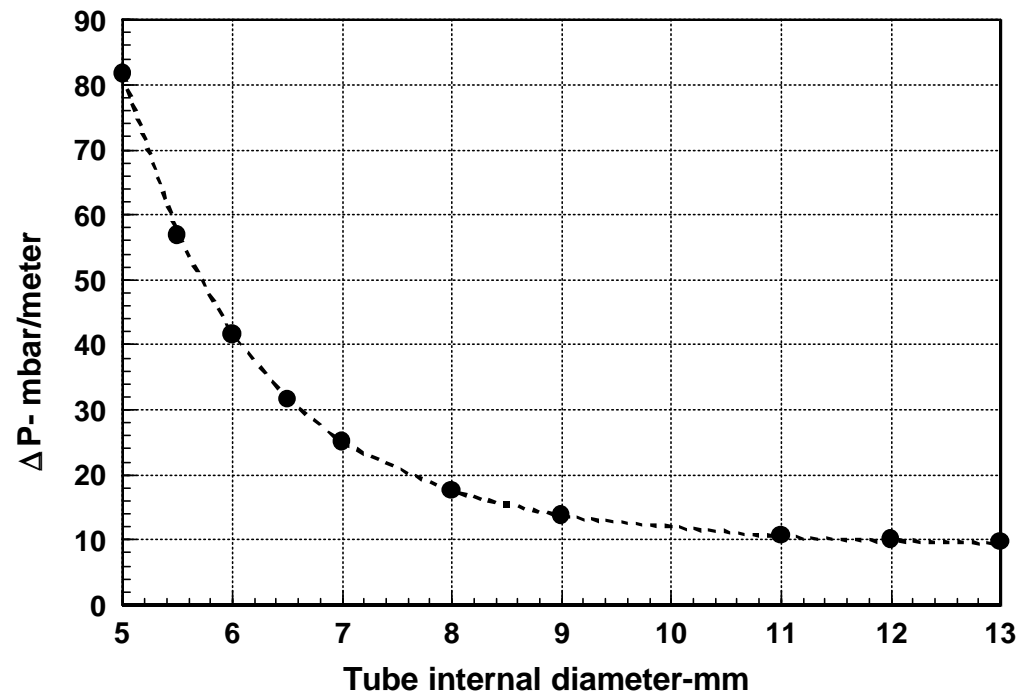
**Goal of >250 mbar not satisfied at exit**  
**(Mach~0.5, compressible flow solution req'd)**

## Two Phase Flow- Pressure Drop (6 mm diameter)



## Two Phase Flow- Line Pressure Drop (88% exit quality)

- Pressure drop contributions
  - two phase flow estimation
  - frictional +gravitational
  - based on separated flow model
- Conclusion
  - Internal tube diameter approaching 7 mm would be recommended
  - pressure drop for 1.5 tube run of 7 mm tube diameter with associated elevation change of 1.5 meter
    - 37.5 mbar, about double to single phase fluid estimation\*



\*slide 10



# Decision on Small Tubes

## Two Phase Flow-versus Single Phase

*<6 mm tube diameter not practical, >6mm desirable*

- Reference P.B. Whalley “Boiling-Condensation and Gas-Liquid Flow” for *two phase flow models*
  - Return pressure loss estimated using separated flow model
  - Frictional and gravitational pressure terms, *no elbow losses included*
  - Void fraction for fluid quality of 0.8 essentially equal to 1, nearly all vapor
- Single Phase Flow
  - Dry vapor return ( $x_o \sim 1.0$ )
  - Compressible flow regime
  - Isothermal flow solution in inner detector region
  - Ignored gravitational term, since density decreased quickly
- Single Phase Results
  - 26.2 mbar/meter first 1.5 meters
  - 35 mbar/meter next 5.4 meters
  - local pressure decayed to point where Mach number approached 0.5
  - iterative solution required
  - Critical flow would occur at tube diameter 4.7 mm and 2.5 meters
  - *Conclusion tube diameter too small, recommend 7 mm initially*
- Two Phase Results
  - Comparable results at quality of 0.98 (singularity occurs at  $\alpha=1$ ) 35 mbar/meter
  - gravitational pressure gradient becomes significant at low quality
  - Pressure loss on the order of 242 mbar in first 6.9 meters at tube diameter of 6 mm, without considering elbow losses
  - *Conclusion tube diameter too small at 6 mm*

# Case A-Summary

## Single Phase- Dry Vapor Return (7 mm diameter return)

Tube Section	Tube Diameter mm	Pressure Loss mbar	Static Exit Pressure mbar	Fluid Velocity m/sec	Fluid Density kg/m <sup>3</sup>	Remarks
@ Stave exit	3.4		450	41.9	5.1	Complete loss in dynamic head
Stave manifold Y-Branch	7 after Branch	8		16.8	5.01	Accounts for pressure loss merging into 7 mm tube
			445.5	12.24	5.05	Entrance to 1.5 m run
1.5 m	7	18.3				
	7	5.8				3 elbows
		24.1 combined	421.4	12.94	4.78	Exit after 1.5 m run
5.4	9	21.2				
		2.3				3 elbows
		23.5 combined	395.7	8.3	4.5	Exit after 5.4 m run
25	13	2.3	394.2	1.7	1.61	Exit after 25 m run

***Tube size resulted in incompressible flow throughout.***

## Two Phase Flow Up to 5.4 meters

Tube Section	Tube Diameter mm	Pressure Loss mbar	Static Exit Pressure mbar	Vapor Velocity m/sec	Pseudo Fluid Density kg/m <sup>3</sup>	Remarks
@ Stave exit	3.4		450	41.9	5.1	Complete loss in dynamic head
Stave manifold Y-Branch	7 after Branch	8				
			442	8.3	7.7	Entrance to 1.5 m run
1.5 m	7	37.5				
	7	11.4				3 elbows
		56.9 combined	385.1	8.3	7.7	Exit after 1.5 m run
5.4	9	74.2		5.7		
		2.3				3 elbows
		76.5 combined	308.6	8.3	4.5	Exit after 5.4 m run
25	13	2.3	306.3*	1.7	1.61	Exit after 25 m run
						*assumed to be all vapor at this point

***More analysis needed--must define at what point system is all vapor  
part of next step in analysis***

## Preliminary Conclusions-Based on Stave Model (7 mm diameter tube)

- Two Phase Flow-estimate
  - *total rough estimate of static pressure 306 mbar*
    - *after 31 meters, with some pressure variance around detector of the order of 8.25 mbar*
- Single Phase Flow-estimate
  - *total rough estimate of static pressure 394 mbar*
    - *after 31 meters, with no known pressure variance around detector of the unless tube geometry is asymmetric*
- Two solution methods give slightly different results
  - *largely the same since tube diameter has been increased*

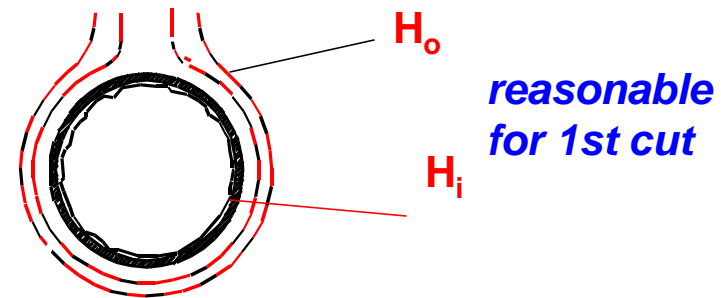
## Vapor Return Tube- Heat Gain

- **Analysis objectives**
  - ultimately establish fluid temperature as function path length
    - point at which dry vapor is attained in the return path, in terms of exit quality
    - fluid density and velocity for updating pressure drop calc's
  - iterate analysis information on tube thermal insulation and thermal boundary conditions become available
    - heat transport influenced by tube bundling arrangements, as well
  - provide information for refrigerant cycle analysis
- **Initial step**
  - solve for free convection heat transfer of isolated tube
    - bound heat gain
  - solve for heat transport for bundled tube arrangement using CFD code as required
    - establish reduction in heat gain from bundled arrangement and confines of walls

## Thermal Boundary and Heat Transfer Coefficient

- **Approach-Initial step**
  - evaluated free convective coefficient for isolated horizontal tube
  - inside film coefficient determined from flow parameters, velocity, etc., from fluid analysis
  - outside film coefficient based conventional method for determining free convection coefficient, i.e.,  $\Delta T$  between surface and surroundings, fluid buoyancy, etc.
  - solution of simultaneous equations
- **More detail needed**
  - *tube bundle arrangement*
    - *adjacent inlet and return lines?*
  - *orientation*
  - *proximity of walls--significant effect on free convection coefficient*

### Isolated horizontal tube



### Multiple tubes bundled, tube shape?

**No evaporation of a fluid,  
based on sensible heat gain only**  
to accurately determine  
at what point residual vapor  
is evaporated is the next step

## Tubing Layout

(latest info courtesy of Eric Anderssen)

Location	Current Diameter	Desired Diameter	Length	Number Exhaust Tubes per side (nominally)		Bundle Dimension for desired tube size		Insulation Thickness	Exterior/Gas Temperature
				Stave	Disk	Inside	Outside		
Pixel Envelope	7	7	0.01	3	3			<i>dry gas</i>	-10
Pixel Envelope	7	7	0.01	3	3			<i>dry gas</i>	-10
Pixel Envelope	7	7	0	3	3			<i>dry gas</i>	-10
Pixel Envelope	7	7	0.15	3	3			<i>dry gas</i>	-10
Pixel Envelope	7	7	0	3	3			<i>dry gas</i>	-10
Pixel Envelope	7	7	0.6	3	3			<i>dry gas</i>	-10
Leave Pixel Envelope	7	7	0.1	3	3			<i>dry gas</i>	-10
SCT Barrel	7	7	0.02	3	3			<i>dry gas</i>	-10
SCT Barrel	7	7	0	3	3			<i>dry gas</i>	-10
SCT Barrel	7	7	0.35	3	3			<i>dry gas</i>	-10
SCT Barrel	7	7	0.02	3	3			<i>dry gas</i>	-10
Leave Thermal Barrier	7	7	0.04	3	3			<i>dry gas</i>	-10
TRT Gap	7	7	0.5	3	3	21 X 14	31 X 24	5mm	20
PPB1	7	7	0	3	3			<i>dry gas</i>	20
PPB1	7 to 7	7 to 9	0.04	3	3			<i>dry gas</i>	20
PPB1	7	9	0	3	3			<i>dry gas</i>	20
Cryostat Bore	7	9	2.5	3	3	27 X 18	38 X 28	5mm	20
PPF1	7	9	0.3	3	3	27 X 18	38 X 28	5mm	20
Cryostat Side	7	9	2	3	3	27 X 18	38 X 28	5mm	25
Enter Tile nooses	7	9	0	3	3	27 X 18	38 X 28	5mm	25
Enter PPB2	7	9	0.2	3	3			<i>dry gas</i>	25
PPB2	7	9	0.3	3	3			<i>dry gas</i>	25
PPB2	7 to 13	9 to 13	0.05	3	3			<i>dry gas</i>	25
PPB2	13	13	0.3	3	3			<i>dry gas</i>	25

Total 2.2 m  
(uninsulated)

1.3 m

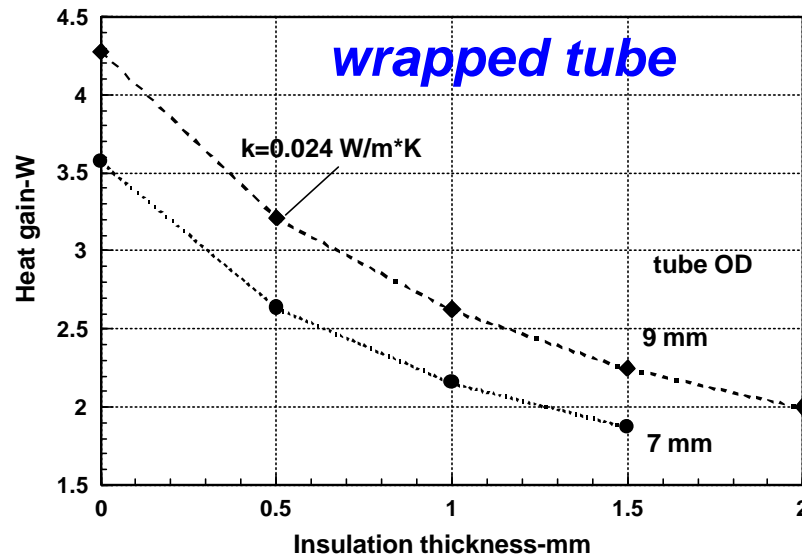
.04 m

.85 m

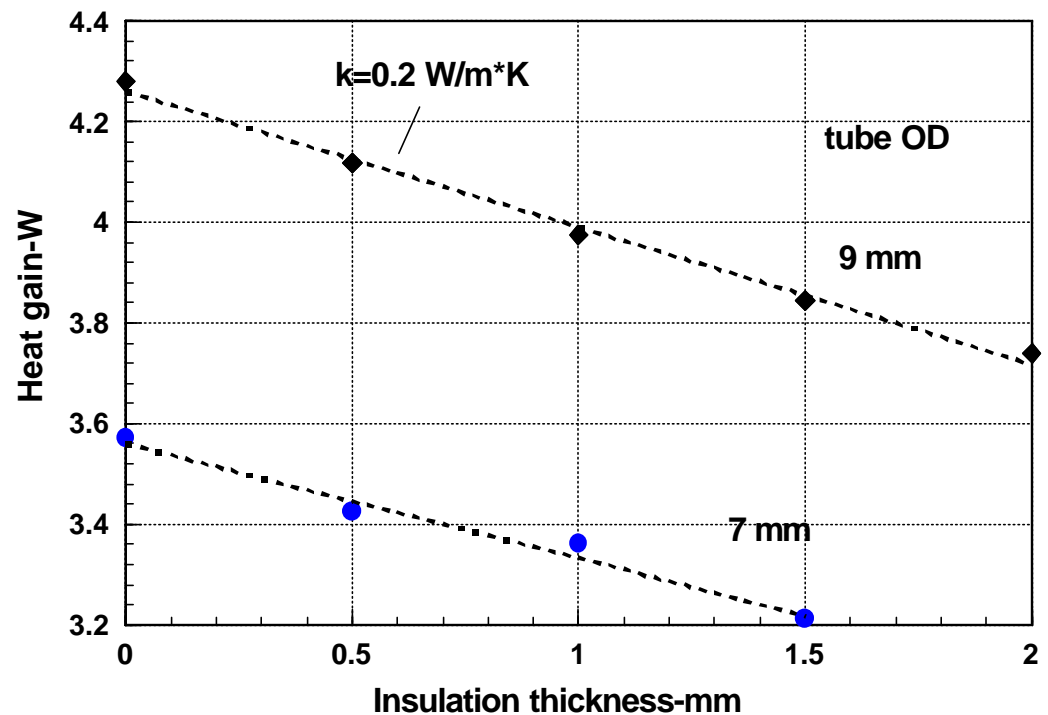
Total path length to this point of 7.5 m

## Tube Heat Gain in Cold Space-Isolated Tube (-10°C)

- *Effect of insulation*
  - wrap tube with simple insulation potentially cuts heat gain
  - creates dead nitrogen gas space



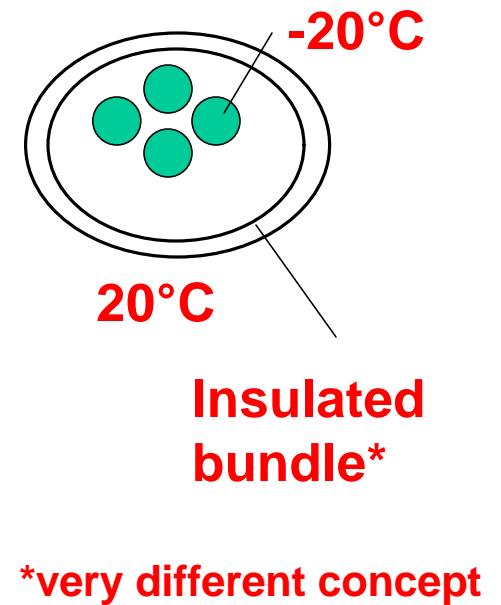
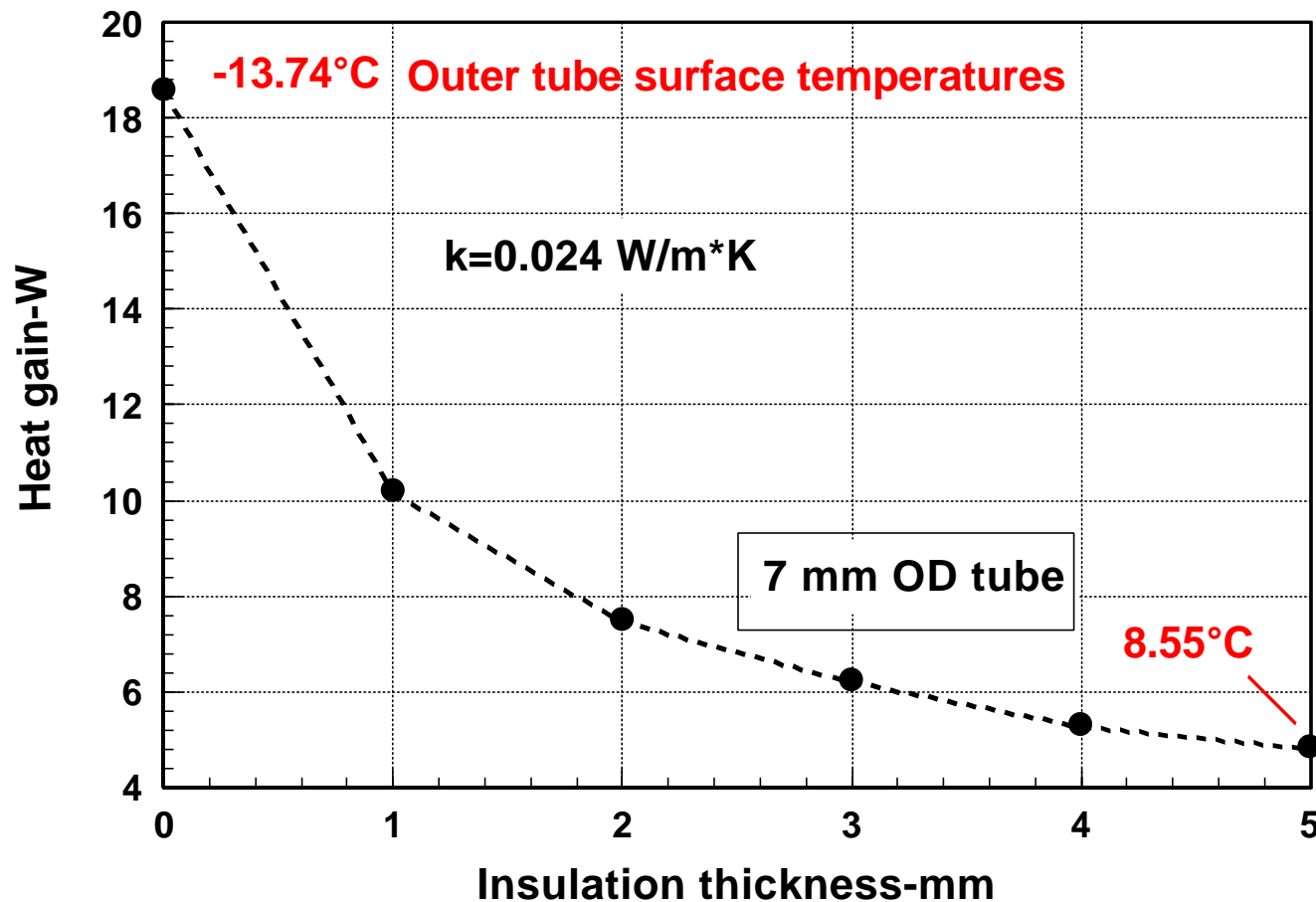
### *polymer coated tube*





# Tube Insulation Analysis

## Tube Heat Gain in Warm Space-*Isolated Tube* (20°C)



## Observations Based on First Cut

- Heat gain in Pixel/SCT cold region (-10°C)
  - Uninsulated, unbundled tube will gain heat via free convection heat transfer
    - unless in close proximity to walls, which disrupts convection
  - An estimate of the sensible heat gain
    - 7 mm OD tube, 4.68 W/m, or 6.1 W in 1.3 meter run
    - Amounts to 3% heat gain within the thermal enclosure @ -10°C
      - based on 202 W, modularity of two
- Heat gain in TRT Gap to PPB2 (20°C)
  - An estimate for same 7 mm, but insulated tube, 4.85 W/m, 5.3 meters, 25.7 W, or 12.7% gain @ 20 °C space temperature
  - Uninsulated portion for 7 mm, 18.8 W/m, 0.89 meters, 16.8 W, or 8.3% @ 20 °C space temperature
    - note tube surface temperature is -13.74°C
- Total vapor return tube heat gain *up to insulated tube region*
  - 48.6 W, or 24%--*may be lower depending on thermal boundary conditions, however*
    - *if totally sensible heat gain--this mounts to 27.6°C increase in vapor temperature*

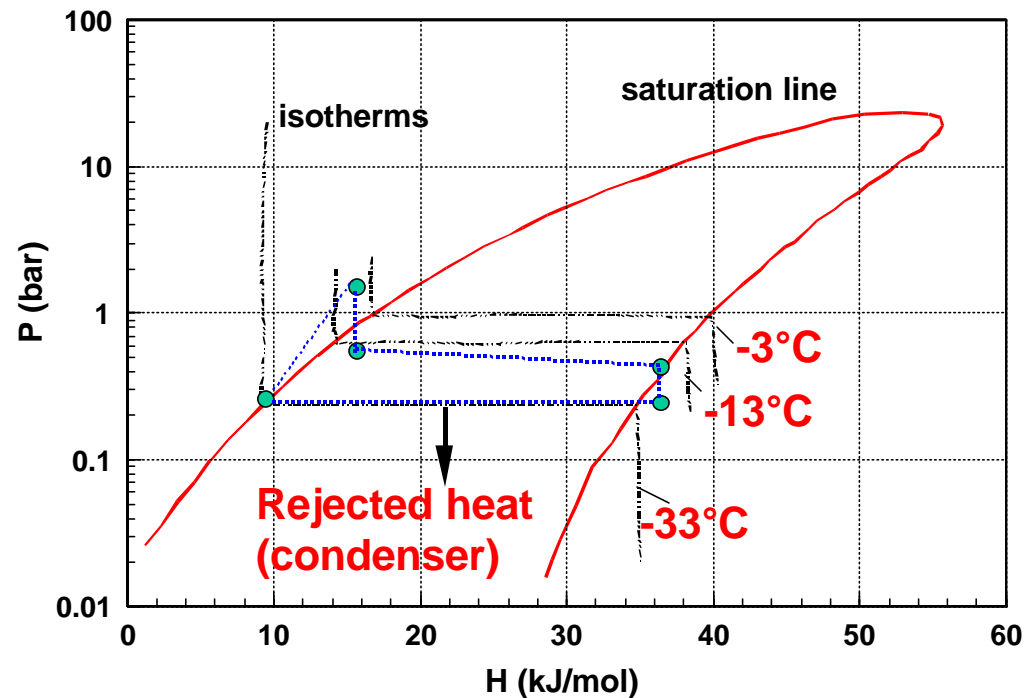
## Comments

- More work is needed, *but*
  - *if fluid exits at a quality of 0.88, i.e., 12% excess cooling capacity per stave, then*
    - fluid can pick-up nominally 12% and remain constant in temperature
      - question is over what distance is required to evaporate
    - mitigating this remark is the extent which the tubes are isolated
      - from pressure drop viewpoint it is desirable to have the fluid evaporate within the first 6 meters
    - if dry vapor exits, the heat transfer solution must be iterated to find the fluid temperature as function of location
      - as temperature increases the inside and outside film coefficients change significantly
  - clearly, a significant effect on predictability exists from the physical constraints
  - Free convective heat transfer coefficients determined for the isolated tube ranged from  $<10$  to  $18 \text{ W/m}^2 \text{ K}$ , which are by most standards quite high

## Objective-Low Pressure ~0.5 bar system

- System issues in low pressure return
  - requirement for 250 mbar minimum pressure at inlet of compressor
  - compounded by distance to compressor
  - need for two stage compression to provide 2 bar inlet pressure
- Condenser approach
  - minimum return pressure limited only by choice of condenser temperature  $T_{sat}$
  - Choice of  $T_{sat}$  influenced by refrigeration power to reject heat back to ambient
  - location close to detector is still important

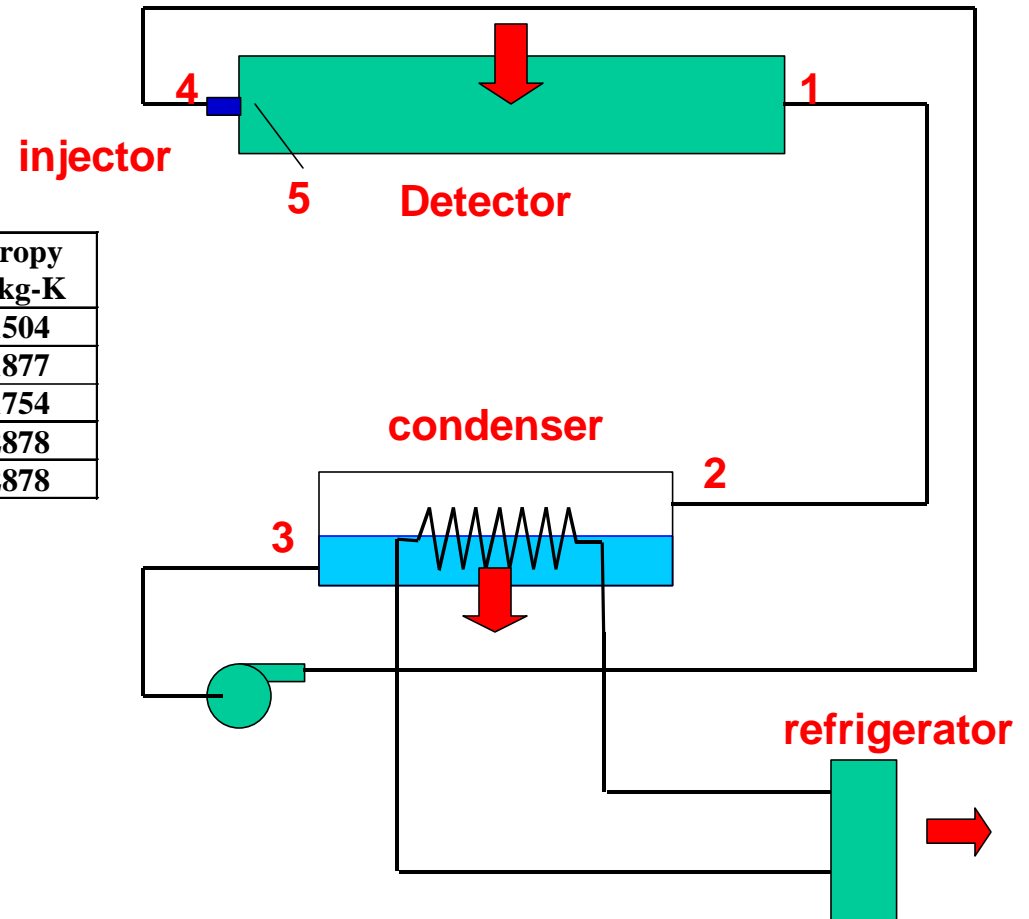
$C_4F_{10}$  Pressure-Enthalpy Curve



## System Schematic

Referenced to T=200K

Point	Pressure mbar	Temperature Degree C	Enthalpy kJ/kg	Entropy KJ/kg-K
1	0.45	-20.5	154.02	0.1504
2	0.238	25	190.521	0.1877
3	.238	-33	38.529	0.1754
4	1.965	-5	67.029	0.2878
5	0.58	-15	67.029	0.2878



## Example Schematic-Arbitrarily Worst Case Scenario

- **Comments**

- No attempt to avoid vapor return temperature reaching 25°C
- Presumes injector temperature of -5°C
- Pressure drop, PT 5 to PT 1, based on two phase calculation for a stove
- Specified pressure of 238 mbar at condenser inlet is arbitrary

- **Results**

- Heat input from, PT 5 to PT 1, is 86.99 kJ/kg
  - if quality  $X_o$  equals 1 at exit, all heat addition is from detector
  - $X_o < 1$ , then some heat is picked-up within the detector space
- Maximum heat gain in return line PT 1 to PT 2 is 36.5 kJ/kg
- Maximum heat input to return liquid to -5 °C, PT 3 to PT 4, 28.5 kJ/kg
- Result forces condenser to remove, PT 2 to PT 3, 151.99 kJ/kg
  - for 15 kW system, condenser rejects 151.99 kJ/kg, or 74.7% more heat than required (11.21 kW excess)
  - optimization of the heat cycle can improve this situation
    - colder inlet to injector
    - thermal isolation of vapor return lines or auxiliary cooling

## What We Would Propose At This Stage-*More Work!*

- Refine cooling cycle analysis
  - predict heat transport in thermostructures
    - establish quality and margin for *dry-out*
    - assess benefit of increasing thermostructure hydraulic diameter for improved heat transport
  - evaluate refrigerator cycle required to pump heat out
    - re-evaluate condenser temperature and return vapor temperature
  - detail analysis of heat transfer associated with vapor and inlet lines to account for heat pick-up and determination of line insulation
    - thermal interaction of cold and warm lines due to their proximity
    - may be more optimum in long tube runs to effectively maintain fluid temperature by secondary cooling loop, e.g., water/methanol
- Full scale experimental mock-up of coolant system
  - Provide semblance of representative operational states for all elements
  - Demonstrate operating parameters for each element
  - compare experimental results with predictions

## Issues Remaining for C<sub>4</sub>F<sub>10</sub> System

- Need to add confidence to the experimental base established to date on the low pressure C<sub>4</sub>F<sub>10</sub> system
  - not clear that a compressor at 140 meters would be acceptable, more analysis is needed
    - need information on required compressor pumping speed
    - results obtained thus far suggest
      - continue investigation of both systems, by adding more detail to the thermal hydraulic analysis
  - refine the proposed condenser concept
    - factor in reality of servicing, maintenance considerations
    - ensure mass accumulation in condenser system will not become a problem
      - centrifugal pump with pressure relief approach
    - condenser sizing
    - need for removing non-condensable gas accumulation?
  - refrigeration system requirements-for condenser concept
    - satisfy heat rejection at -33 °C
    - evaluate option of higher condenser temperature