# Reference: Boiling and Condensation and Gas-Liquid Flow, Whalley

# Frictional Pressure Drop Analysis-Stave Barrel C3F8, Quality of 5% after injection using Friedel correlation

mbar :=  $10^{-3}$ bar uPa :=  $10^{-6}$ Pa kJ := 1000J t := .012in O := 240W L1 := 2m tube length, round trip  $c_h := 4.9 \text{mm} - 2 \cdot t$   $c_r := \frac{4.9}{2} \text{mm} - t$   $w_c := 2 \text{mm}$   $A_c := w_c \cdot c_h + \pi \cdot c_r^2$  $P_c := 2 \cdot w_c + 2 \cdot \pi \cdot c_r$   $D_h := 4 \cdot \frac{A_c}{P}$   $D_h = 5.272 \cdot mm$   $A_t := A_c$ 

## C3F8 Fluid Properties at -25C

 $T_i := (273.15 - 25)K$   $T_i = 248.15K$   $\mu_V := 10.28\mu Pa \cdot s$  $\mu_{lig} := 267.5 \mu Pa \cdot s$  $c_{liq} \coloneqq 1019 \frac{J}{kg \cdot K} \qquad \rho_{liq} \coloneqq 1565 \cdot \frac{kg}{m^3} \qquad \rho_{V} \coloneqq 16.39 \frac{kg}{m^3}$  $k_{v} \coloneqq 0.009 \frac{W}{m \cdot K} \qquad k_{liq} \coloneqq 0.053 \frac{W}{m \cdot K} \qquad \sigma \coloneqq .015 \frac{N}{m} \qquad \qquad \text{perfluoropropane surface tension at 253.15K}$  $\mathbf{h}_{\text{liq}} \coloneqq 173.7 \frac{\text{kJ}}{\text{kg}} \qquad \mathbf{h}_{\text{v}} \coloneqq 275.6 \frac{\text{kJ}}{\text{kg}} \qquad \Delta \mathbf{h} \coloneqq \mathbf{h}_{\text{v}} - \mathbf{h}_{\text{liq}} \qquad \Delta \mathbf{h} = 101.9 \cdot \frac{\text{kJ}}{\text{kg}} \qquad \lambda \coloneqq \Delta \mathbf{h}$ 

# Tube Fluid Parameters, based on inlet and exit flow quality

$$\mathbf{x} \coloneqq .05 \qquad \mathbf{x}_{\mathbf{0}} \coloneqq .85 \qquad \text{mdot} \coloneqq \frac{\mathbf{Q}}{\left(\mathbf{x}_{\mathbf{0}} - \mathbf{x}\right) \cdot \Delta \mathbf{h}} \qquad \text{mdot} = 2.944 \times 10^{-3} \frac{\mathrm{kg}}{\mathrm{s}} \qquad \nu_{\mathrm{liq}} \coloneqq \frac{\mu_{\mathrm{liq}}}{\rho_{\mathrm{liq}}}$$

$$G_{liq} := mdot \cdot (1 - x)$$
  $G_{liq} = 2.797 \times 10^{-3} \frac{kg}{s}$   $G_{v} := mdot - G_{liq}$   $G_{v} = 1.472 \times 10^{-4} \frac{kg}{s}$ 

$$G_t := \frac{\text{mdot}}{A_t}$$
  $R_{\text{liq}} := \frac{G_t \cdot D_h}{\mu_{\text{liq}}}$   $R_{\text{liq}} = 2.519 \times 10^3$ 

$$R_{fgo} := \frac{G_{t} D_{h}}{\mu_{v}}$$
  $R_{fgo} = 6.554 \times 10^{4}$   $C_{fgo} := 0.079 \cdot R_{fgo}^{-0.25}$   $C_{fgo} = 4.937 \times 10^{-3}$ 

$$R_{flo} := \frac{G_t \cdot D_h}{\mu_{liq}}$$
  $R_{flo} = 2.519 \times 10^3$   $C_{flo} := 0.079 \cdot R_{flo}^{-0.25}$   $C_{flo} = 0.011$ 

$$\Phi 2^{2} = E + \frac{3.24 \cdot F2 \cdot Hf}{(Fr)^{0.045} \cdot (We)^{0.035}}$$
 basic equation for two phase flow correction to single phase flow pressure drop

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a1 := 
$$\frac{\rho_{\text{liq}} \cdot C_{\text{fgo}}}{\rho_{\text{v}} \cdot C_{\text{flo}}}$$
 a1 = 42.277 b1 :=  $\frac{G_t^2 \cdot D_h}{\sigma}$  d1 :=  $\frac{G_t^2}{g \cdot D_h}$ 

$$Hf := \left(\frac{\rho_{liq}}{\rho_{v}}\right)^{0.91} \cdot \left(\frac{\mu_{v}}{\mu_{liq}}\right)^{0.19} \cdot \left(1 - \frac{\mu_{v}}{\mu_{liq}}\right)^{0.7} \qquad Hf = 33.183$$

We := 
$$b1 \cdot \left(\frac{x}{\rho_V} + \frac{1-x}{\rho_{liq}}\right)$$
 Fr :=  $d1 \cdot \left(\frac{x}{\rho_V} + \frac{1-x}{\rho_{liq}}\right)^2$  F2 :=  $x^{0.78} \cdot (1-x)^{0.224}$  E :=  $(1-x)^2 + x^2 \cdot a1$ 

$$z := (1 - x)^{2} + x^{2} \cdot (a1) + \frac{3.24 \cdot \left[x^{0.78} \cdot (1 - x)^{0.224}\right] \cdot Hf}{\left[d1 \cdot \left(\frac{x}{\rho_{V}} + \frac{1 - x}{\rho_{liq}}\right)^{2}\right]^{0.045} \cdot \left[b1 \cdot \left(\frac{x}{\rho_{V}} + \frac{1 - x}{\rho_{liq}}\right)\right]^{0.035}}$$
 to be integrated over tube length

 $dpdz_{lo} \coloneqq \frac{2 \cdot C_{flo} \cdot G_t^2}{D_h \cdot \rho_{liq}}$ 

frictional pressure drop based on fluid being solely single phase

$$\Delta P_{f} := dpdz_{lo} \cdot \frac{L1}{0.85 - 0.05} \cdot \int_{0.05}^{0.85} (1 - x)^{2} + x^{2} \cdot (a1) + \frac{3.24 \cdot \left[x^{0.78} \cdot (1 - x)^{0.224}\right] \cdot Hf}{\left[d1 \cdot \left(\frac{x}{\rho_{V}} + \frac{1 - x}{\rho_{liq}}\right)^{2}\right]^{0.045} \cdot \left[b1 \cdot \left(\frac{x}{\rho_{V}} + \frac{1 - x}{\rho_{liq}}\right)^{2}\right]^{0.035}} dx$$

$$\Delta P_{f} = 3.639 \times 10^{3} \cdot Pa$$
  $\Delta P_{f} = 36.394 \cdot mbar$ 

$$\int_{0.05}^{0.85} (1-x)^2 + x^2 \cdot (a1) + \frac{3.24 \cdot \left[x^{0.78} \cdot (1-x)^{0.224}\right] \cdot Hf}{\left[d1 \cdot \left(\frac{x}{\rho_V} + \frac{1-x}{\rho_{liq}}\right)^2\right]^{0.045} \cdot \left[b1 \cdot \left(\frac{x}{\rho_V} + \frac{1-x}{\rho_{liq}}\right)\right]^{0.035}} dx = 32.978$$

this is the correction for the two phase flow, as a multiplier

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$$\Delta P_{lo} \coloneqq 2 \cdot \frac{C_{flo} \cdot G_t^2}{D_h \cdot \left(\frac{x}{\rho_V} + \frac{1 - x}{\rho_{liq}}\right)^{-1}} \cdot L1 \qquad \Delta P_{lo} = 5.054 \cdot mbar$$

$$\Delta P := \Delta P_{lo} + \Delta P_{f} \qquad \Delta P = 41.447 \cdot \text{mbar}$$

# Reference: Evaporative Cooling-Conceptual Design for ATLAS SCT, T.O. Niinikoski

Pressure drop due to momentum transfer , inlet to outlet,  $\Delta P_m = \Phi_m m dot^2 / (At^2 \rho_{lig})$ 

$$x_{in} := 0.05$$
  $x_{out} := 0.85$ 

$$\rho_{hi} := \left(\frac{x_{in}}{\rho_{v}} + \frac{1 - x_{in}}{\rho_{liq}}\right)^{-1} \quad \text{homogeneous flow density} \qquad \rho_{hi} = 273.398 \frac{kg}{m^{3}} \qquad \text{at inlet}$$

$$\frac{\rho_{liq}}{\rho_{hi}} = 5.724 \qquad \frac{\rho_{v}}{\rho_{hi}} = 0.06 \qquad \text{volume fraction of constituents at the inlet}$$

$$\left(\frac{x_{out}}{\rho_{hi}} + \frac{1 - x_{out}}{\rho_{hi}}\right)^{-1} \qquad \text{transformed at the inlet}$$

$$\rho_{ho} := \left(\frac{x_{out}}{\rho_v} + \frac{1 - x_{out}}{\rho_{liq}}\right)^{-1} \quad \text{homogeneous flow density} \qquad \rho_{ho} = 19.247 \frac{\text{kg}}{\text{m}^3} \qquad \text{at outlet}$$

Relative volume fraction of liquid and vapor phases at the inlet and exit

$$j_{in} := \frac{x_{in} \cdot mdot}{\rho_V \cdot A_t} + \frac{\left(1 - x_{in}\right) \cdot mdot}{\rho_{liq} \cdot A_t} \qquad \qquad j_{in} = 0.467 \frac{m}{s} \qquad \qquad \text{total volume flux at inlet}$$

$$j_{O} := \frac{x_{O} \cdot mdot}{\rho_{V} \cdot A_{t}} + \frac{(1 - x_{O}) \cdot mdot}{\rho_{liq} \cdot A_{t}} \qquad \qquad j_{O} = 6.64 \frac{m}{s} \qquad \qquad \text{total volume flux at exit}$$

$$j_{vin} := \frac{x_{in}}{x_{in} + (1 - x_{in}) \cdot \frac{\rho_v}{\rho_{liq}}} \qquad \qquad j_{vin} = 0.834 \qquad \text{volume fraction of vapor at inlet}$$

$$j_{liqin} \coloneqq 1 - j_{vin} \qquad j_{liqin} = 0.166 \qquad \text{volume fraction of liquid at inlet}$$

$$j_{vo} \coloneqq \frac{x_o}{x_o + (1 - x_o) \cdot \frac{\rho_v}{\rho_{liq}}} \qquad j_{vo} = 0.998 \qquad \text{volume fraction of vapor at inlet}$$

$$j_{VO} = 0.998$$
 volume fraction of vapor at inlet

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$$j_{liqo} \coloneqq 1 - j_{vo} \qquad \qquad j_{liqo} = 1.845 \times 10^{-3} \text{volume fraction of liquid at inlet}$$

$$\Phi_m \coloneqq \frac{\left(1 - x_o\right)^2}{j_{liqo}} - \frac{\left(1 - x_{in}\right)^2}{j_{liqin}} + \left(\frac{x_o^2}{j_{vo}} - \frac{x_{in}^2}{j_{vin}}\right) \frac{\rho_{liq}}{\rho_v} \qquad \Phi_m = 75.588$$

Pressure difference to momentum change  $\Delta P_m$ 

$$\Delta P_{m} \coloneqq \Phi_{m} \cdot \frac{\text{mdot}^{2}}{A_{t}^{2} \cdot \rho_{liq}} \qquad \Delta P_{m} = 7.888 \cdot \text{mbar}$$

 $\label{eq:problem_transform} \mbox{Total Pressure Drop} \qquad \Delta P_T := \Delta P + \Delta P_m \qquad \Delta P_T = 49.335 \cdot mbar$ 

## A new source for predicting pressure drop

# R. Reinhard, Y. Hwang Vapor Compression Heat Pumps with Refrigerant Mixtures. Martinelli and Nelson Correlation

 $C_{fgo}$  agrees with their  $f_{fo}$  and since dpdz<sub>lo</sub> = their same term

$$\Delta P_{\text{fnew}} := dpdz_{\text{lo}} \cdot \frac{L1}{(0.85 - 0.05)} \cdot \int_{0.05}^{0.85} \left(1 + \frac{1}{x^{0.5}}\right)^4 \cdot (1 - x)^{1.75} dx$$

$$\Delta P_{\text{fnew}} = 5.111 \times 10^3 Pa$$
  $\Delta PT\text{new} := \Delta P_{\text{fnew}} + \Delta P_m$ 

 $\Delta PTnew = 5.899 \times 10^{3} Pa$   $\Delta PTnew = 58.994 \cdot mbar$  with acceleration pressure drop

#### **Change in Temparture**

The change in saturation temperature corresponding to the pressure drop is as follows:

use higher of two methods and use the change in  $\Delta T/\Delta P$ 

at -25C the  $\Delta T/\Delta P$ = K/6800Pa

$$\Delta T_{sat} := \Delta PT_{new} \frac{K}{6800Pa}$$
 use 59mbar for Table 4  
$$\Delta T_{sat} = 0.868 K$$
 use 1.0C for Table 4